

# **Fluid Mechanics**

Chapter 4:

## **Analytical solutions to the Navier-Stokes equations**

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# Analytic solutions to the Navier-Stokes equations

Some illustrative Examples

A. Couette flow

B. Poiseuille flow

C. Flow in a circular pipe

D. Flow near an Oscillating Plate

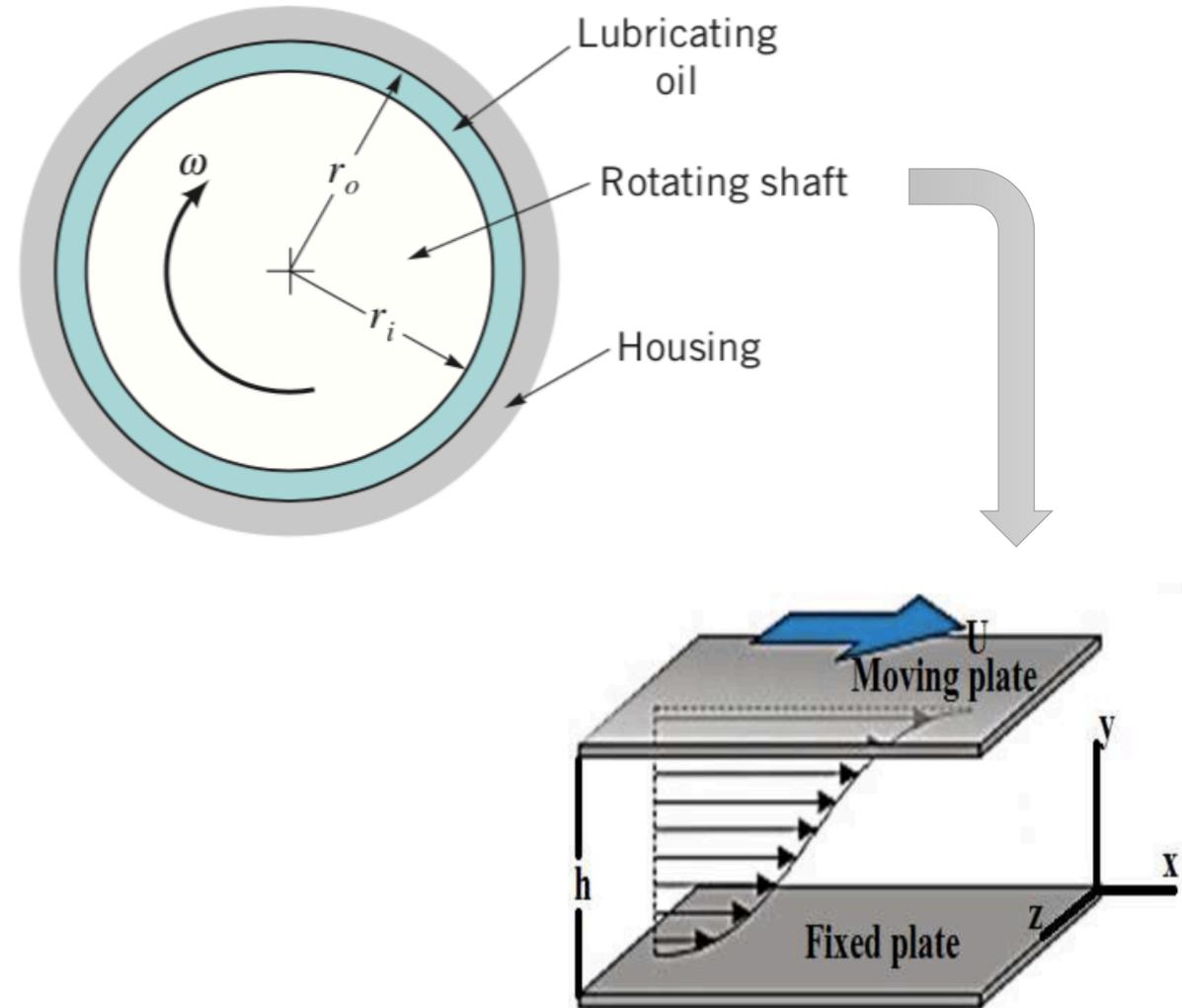
## A. Flow between a Fixed and a Moving Plate: Plane Couette Flow

### Applications:

- Lubrication
- Geological systems
- Painting, cleaning etc. (thin-film applications)

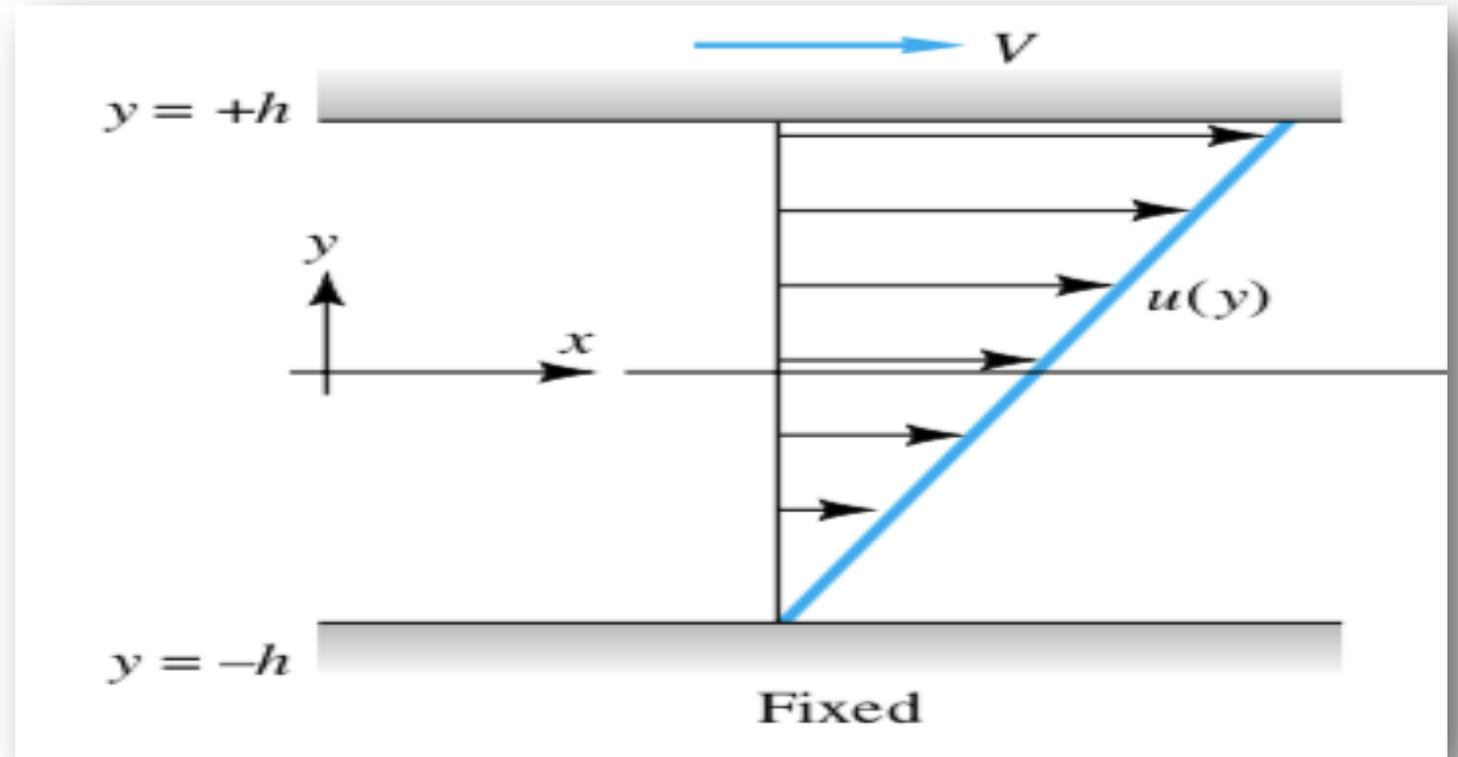


**Journal Bearing**



## A. Flow between a Fixed and a Moving Plate: Plane Couette Flow

- Two-dimensional
- Incompressible
- Plane
- Viscous flow
- Between parallel plates a distance  $2h$  apart
- Assume that the plates are very wide and very long, so that the flow is essentially axial
- The upper plate moves at velocity  $V$  but there is no pressure gradient
- Neglect gravity effects



The continuity equation

$$\frac{\partial u}{\partial x} + \frac{\partial v}{\partial y} + \frac{\partial w}{\partial z} = 0 = \frac{\partial u}{\partial x} + 0 + 0 \quad \text{or} \quad u = u(y) \text{ only}$$

## A. Flow between a Fixed and a Moving Plate: Plane Couette Flow

- Thus there is a single nonzero axial-velocity component which varies only across the channel.
- The flow is said to be fully developed (far downstream of the entrance).
- The Navier-Stokes momentum equation for two-dimensional (x, y) flow:

$$\rho \left( u \frac{\partial u}{\partial x} + v \frac{\partial u}{\partial y} \right) = -\frac{\partial p}{\partial x} + \rho g_x + \mu \left( \frac{\partial^2 u}{\partial x^2} + \frac{\partial^2 u}{\partial y^2} \right)$$

$$\rho(0 + 0) = 0 + 0 + \mu \left( 0 + \frac{d^2 u}{dy^2} \right)$$

$$\frac{d^2 u}{dy^2} = 0 \quad \text{or} \quad u = C_1 y + C_2$$

The no-slip condition

At  $y = +h$ :

$$u = V = C_1 h + C_2$$

At  $y = -h$ :

$$u = 0 = C_1(-h) + C_2$$

$$C_1 = \frac{V}{2h} \quad \text{and} \quad C_2 = \frac{V}{2}$$

## A. Flow between a Fixed and a Moving Plate: Plane Couette Flow

The solution for the flow between plates with a moving upper wall, is:

$$u = \frac{V}{2h} y + \frac{V}{2} \quad -h \leq y \leq +h$$

This is *Couette flow* due to a moving wall: a linear velocity profile with no-slip at each wall, as anticipated.

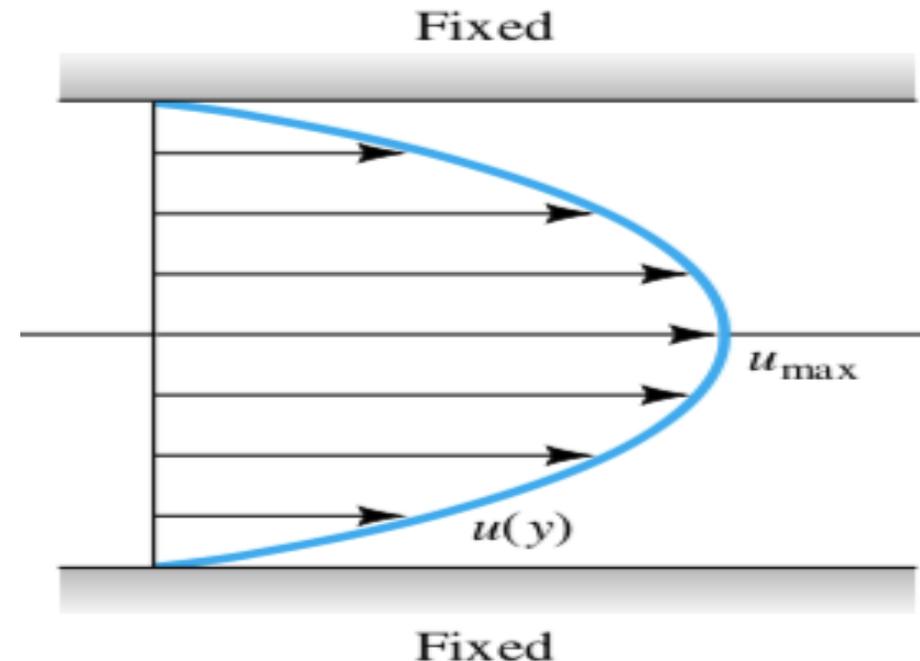
## B. Flow due to Pressure Gradient between Two Fixed Plate: Plane Poiseuille Flow (Channel Flow)

- Poiseuille flows are driven by pumps that forces the fluid to flow by modifying the pressure.
- Fluids flow naturally from regions of high pressure to regions of low pressure.
- Typical examples are cylindrical pipe flow and other duct flows.
- Figure below illustrates a fully developed plane channel flow.
- Fully developed Poiseuille flows exists only far from the entrances and exits of the ducts, where the flow is aligned parallel to the duct walls.

### Assumptions:

- Both plates are fixed.
- The pressure varies in the  $x$  direction.
- The gravity is neglected.
- The  $x$ -momentum equation changes only because the pressure is variable:

$$\mu \frac{d^2 u}{dy^2} = \frac{\partial p}{\partial x}$$



## B. Flow due to Pressure Gradient between Two Fixed Plate: Plane Poiseuille Flow (Channel Flow)

The  $y$ - and  $z$ -momentum equations lead to:  $\frac{\partial p}{\partial y} = 0$  and  $\frac{\partial p}{\partial z} = 0$  or  $p = p(x)$  only

Thus the pressure gradient is the total and only gradient:  $\mu \frac{d^2 u}{dy^2} = \frac{dp}{dx} = \text{const} < 0$

The solution is accomplished by double integration:  $u = \frac{1}{\mu} \frac{dp}{dx} \frac{y^2}{2} + C_1 y + C_2$

The constants are found from the no-slip condition at each wall:

$$\text{At } y = \pm h: \quad u = 0 \quad \text{or} \quad C_1 = 0 \quad \text{and} \quad C_2 = -\frac{dp}{dx} \frac{h^2}{2\mu}$$

## B. Flow due to Pressure Gradient between Two Fixed Plate: Plane Poiseuille Flow (Channel Flow)

Thus the solution to the flow in a channel due to pressure gradient, is:

$$u = -\frac{dp}{dx} \frac{h^2}{2\mu} \left(1 - \frac{y^2}{h^2}\right)$$

The flow forms a *Poiseuille* parabola of constant negative curvature. The maximum velocity occurs at the centerline:

$$u_{\max} = -\frac{dp}{dx} \frac{h^2}{2\mu}$$

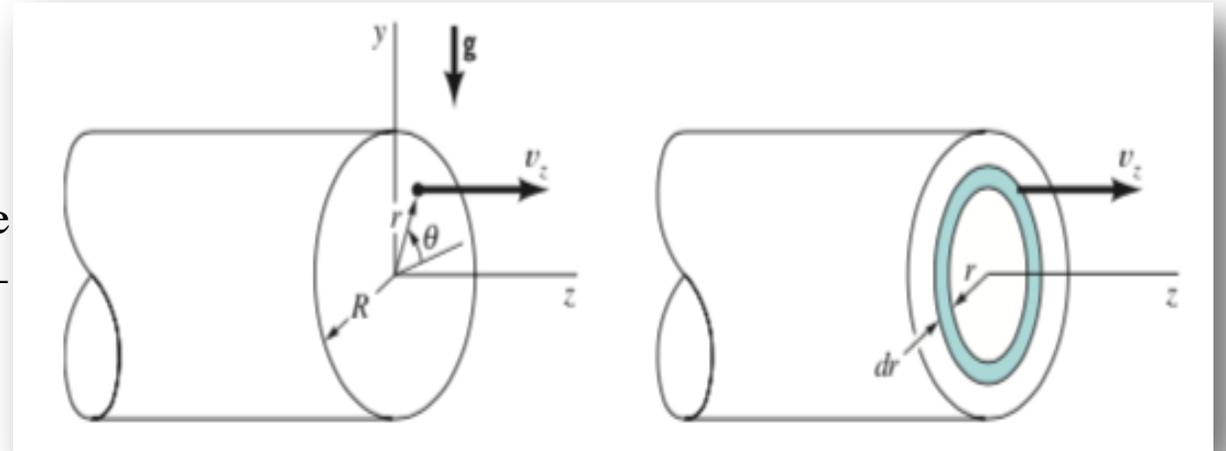
## C. Flow in circular pipe

If we assume the flow is in the x-direction, then the velocity vector is  $u(y, z, t) = (u(y, z, t), 0, 0)$  and Navier-Stokes equations can be simplified to

$$\frac{\partial u}{\partial t} = \nu \nabla^2 u - \frac{1}{\rho} \frac{dp}{dx},$$

For a steady cylindrical pipe flow with radius  $r_0$  the solution is found simply by integrating twice and applying boundary conditions:

$$\begin{aligned} \frac{\nu}{r} \frac{d}{dr} \left( r \frac{du}{dr} \right) &= \frac{1}{\rho} \frac{dp}{dx}, \\ \frac{d}{dr} \left( r \frac{du}{dr} \right) &= \frac{r}{\mu} \frac{dp}{dx}, \\ r \frac{du}{dr} &= \frac{r^2}{2\mu} \frac{dp}{dx} + C_1, \\ \frac{du}{dr} &= \frac{r}{2\mu} \frac{dp}{dx} + \frac{C_1}{r}, \\ u(r) &= \frac{r^2}{4\mu} \frac{dp}{dx} + C_1 \ln(r) + C_2, \end{aligned}$$



## C. Flow in circular pipe

where  $C_1$  and  $C_2$  are integration constants. The constant  $C_1$  must be zero for the flow at the center of the pipe to remain finite. The condition at the wall  $u(r_0) = 0$  gives

$$C_2 = -\frac{r_0^2}{4\mu} \frac{\partial p}{\partial x},$$

which leads to the final expression for the pipe flow

$$u(r) = -\frac{1}{4\mu} \frac{\partial p}{\partial x} (r_0^2 - r^2)$$

## D. Flow near an Oscillating Plate

Consider that special case of a viscous fluid near a wall that is set suddenly in motion as shown in Figure 1. The unsteady Navier-Stokes reduces to:

$$\frac{\partial u}{\partial t} = \nu \frac{\partial^2 u}{\partial y^2}$$

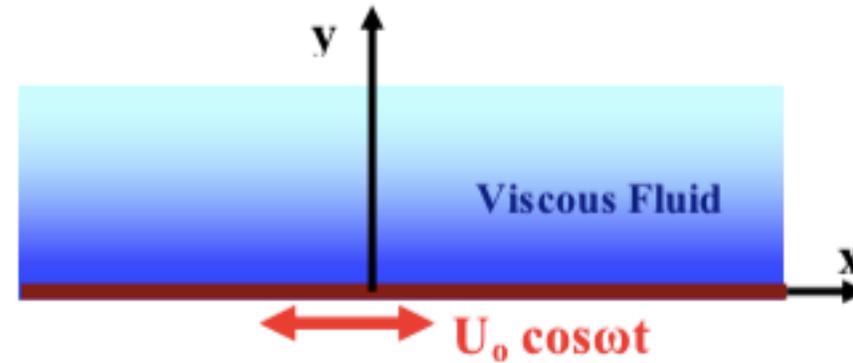


Figure 2. Schematics of flow near an oscillating wall.

The boundary conditions are:

$$u = U_0 \cos \omega t \quad \text{at } y = 0$$

$$u = 0 \quad \text{at } y = \infty$$

## D. Flow near an Oscillating Plate

Let

$$u = U_0 e^{-ky} \cos(\omega t - ay).$$

Then

$$\frac{\partial u}{\partial t} = -\omega U_0 e^{-ky} \sin(\omega t - ay)$$

$$\frac{\partial u}{\partial y} = U_0 e^{-ky} (-k \cos(\omega t - ay) + a \sin(\omega t - ay))$$

$$\frac{\partial^2 u}{\partial y^2} = U_0 e^{-ky} (k^2 \cos \theta - 2ka \sin \theta - a^2 \cos \theta), \quad \theta = \omega t - ay$$

## D. Flow near an Oscillating Plate

Substituting the last two equations into the reduced Navier-stokes equation, it follows that

$$-\omega \sin \theta = \nu \left( (k^2 - a^2) \cos \theta - 2ak \sin \theta \right)$$

$$a^2 = k^2$$

$$\omega = 2ak\nu = 2k^2\nu$$

$$k = \sqrt{\frac{\omega}{2\nu}} = a$$

Thus, the velocity profile is given as

$$u = U_0 e^{-ky} \cos(\omega t - ky), \quad k = \sqrt{\frac{\omega}{2\nu}}.$$