

PEF-5750  
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*Discrete Systems:  
ARGYRIS' NATURAL MEMBRANE ELEMENT*

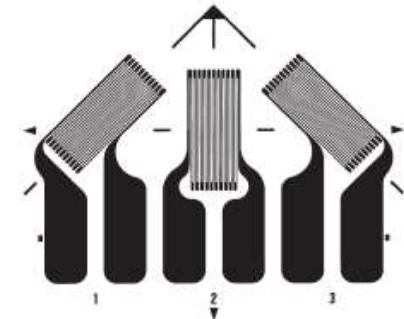
*08/11/2018*

# Argyris' Natural Membrane Element

Argyris ~1974

A membrane finite element based on natural deformations  
(measured along the sides of the element),  
able to cope with large displacements and large deformations.

- Akin to a “strain rosette” plane stress finite element:



Meek ~1991

- A corrotational description;
- Small strains.

Pauletti ~2003

- a more concise notation;
- distinction between the constitutive and geometric parts of the element tangent stiffness;
- the “simplest possible membrane finite element”;
- large displacements / small strains (a few percent...)

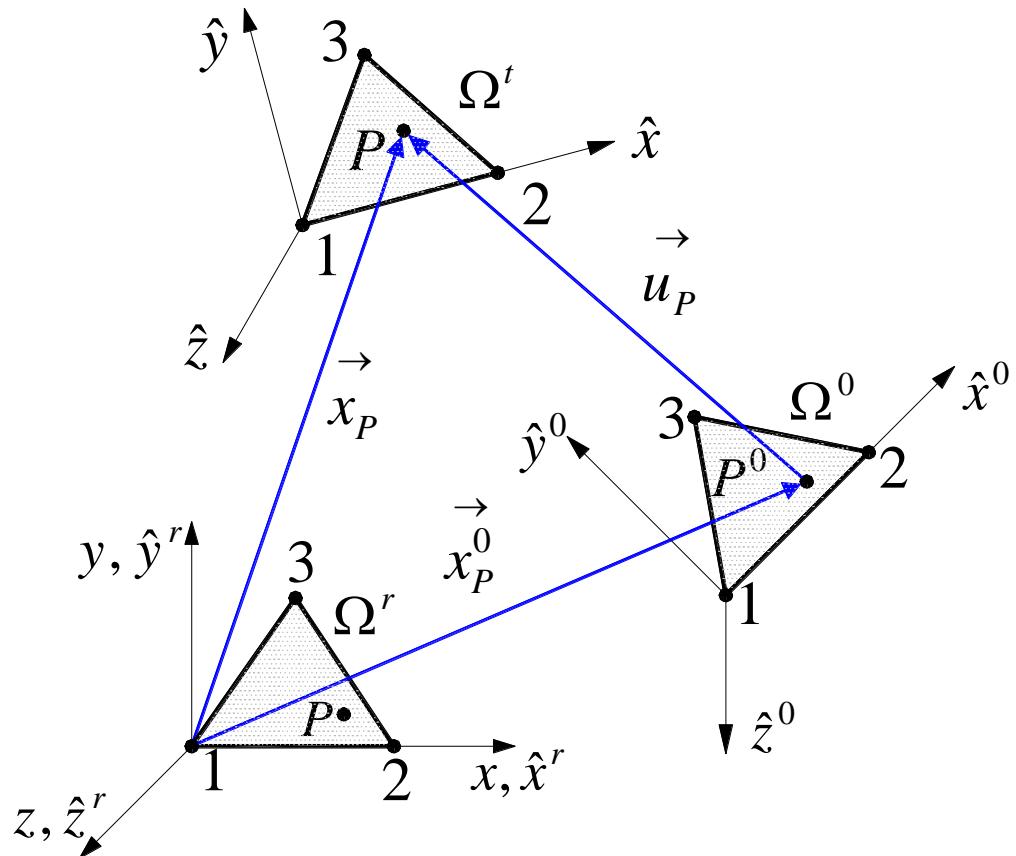
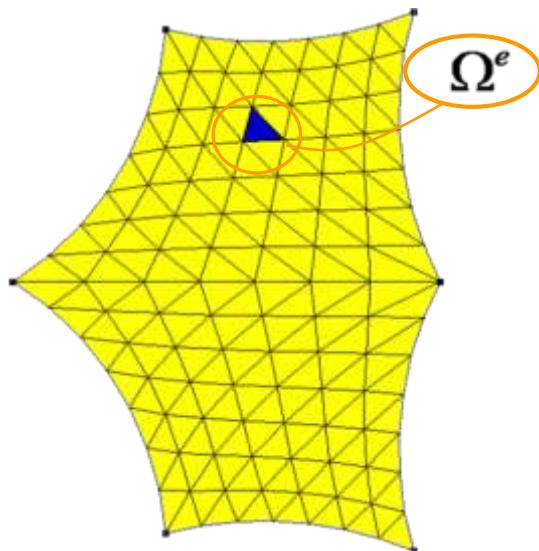
Pauletti (2006)

- first publication on the natural force density concept

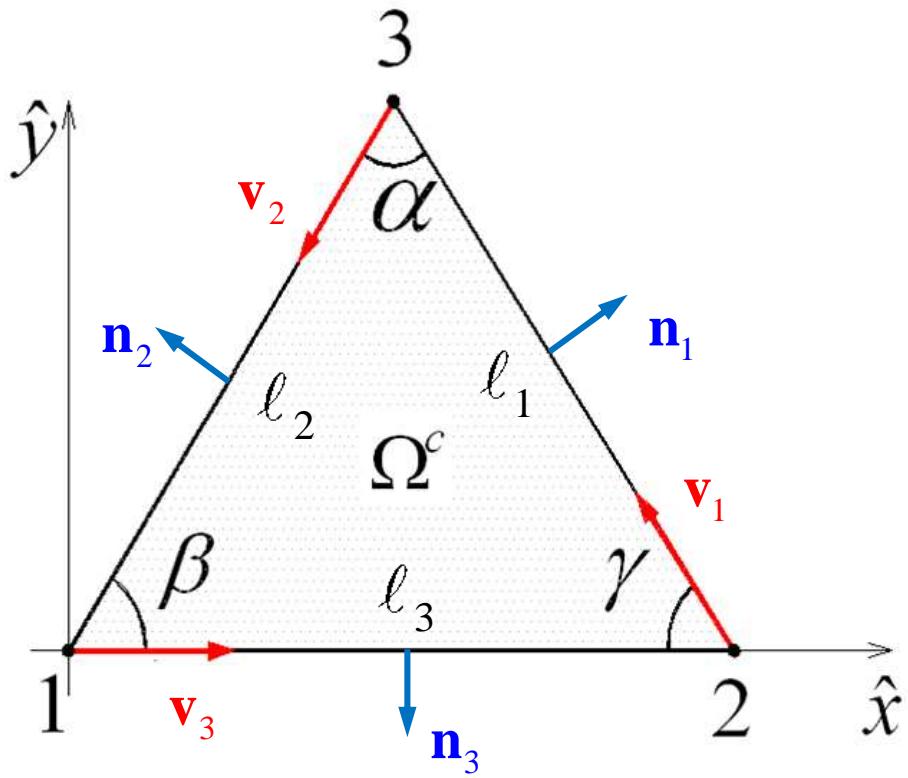
R.M.O. Pauletti, “An extension of the force density procedure to membrane structures”

IASS Symposium / APCS Conference – New Olympics, New Shell and Spacial Structures, Beijing, 2006

# Reference, Initial and Current Configurations For Argyris Element



# Element Description



$$\mathbf{x}_{0,c} = \begin{bmatrix} \mathbf{x}_1 \\ \mathbf{x}_2 \\ \mathbf{x}_3 \end{bmatrix}_{0,c}$$

$$\mathbf{u} = \mathbf{x}_c - \mathbf{x}_0 = \begin{bmatrix} \mathbf{u}_1 \\ \mathbf{u}_2 \\ \mathbf{u}_3 \end{bmatrix}$$

Side lengths:

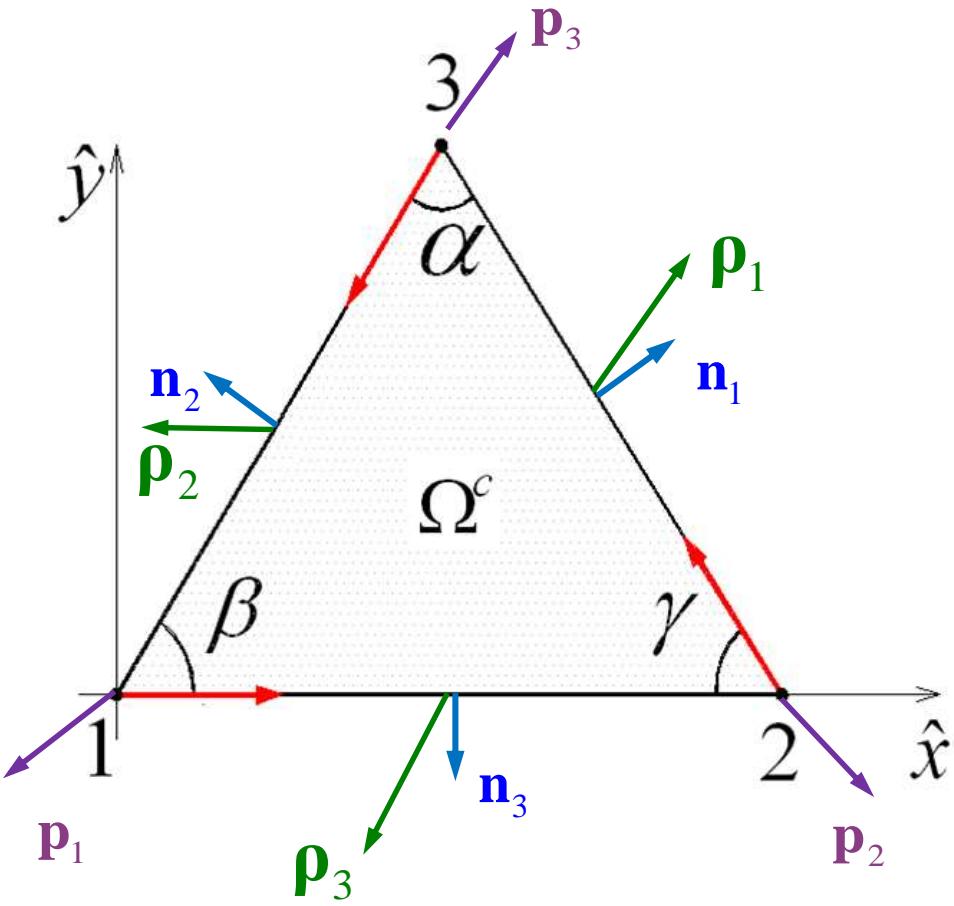
$$\ell_i = \|\mathbf{x}_k - \mathbf{x}_j\| \quad \{i, j, k\} \equiv \{1, 2, 3\} \\ (\text{in cyclic permutation})$$

Unit side vectors:

$$\mathbf{v}_i = \frac{\mathbf{x}_k^e - \mathbf{x}_j^e}{\ell_i}$$

Unit normal vectors:  $\mathbf{n}_i = -\hat{\mathbf{k}} \times \mathbf{v}_i$

# Element Stress Field and Vector of Internal Nodal Forces



Cauchy Plane Stress Tensor:

$$\hat{\boldsymbol{\sigma}} = \begin{bmatrix} \sigma_{\hat{x}} & \tau_{\hat{x}\hat{y}} & 0 \\ \tau_{\hat{x}\hat{y}} & \sigma_{\hat{y}} & 0 \\ 0 & 0 & 0 \end{bmatrix} \equiv \begin{bmatrix} \sigma_{\hat{x}} \\ \sigma_{\hat{y}} \\ \tau_{\hat{x}\hat{y}} \end{bmatrix}$$

Side stress vectors:

$$\mathbf{p}_i = \hat{\boldsymbol{\sigma}} \mathbf{n}_i$$

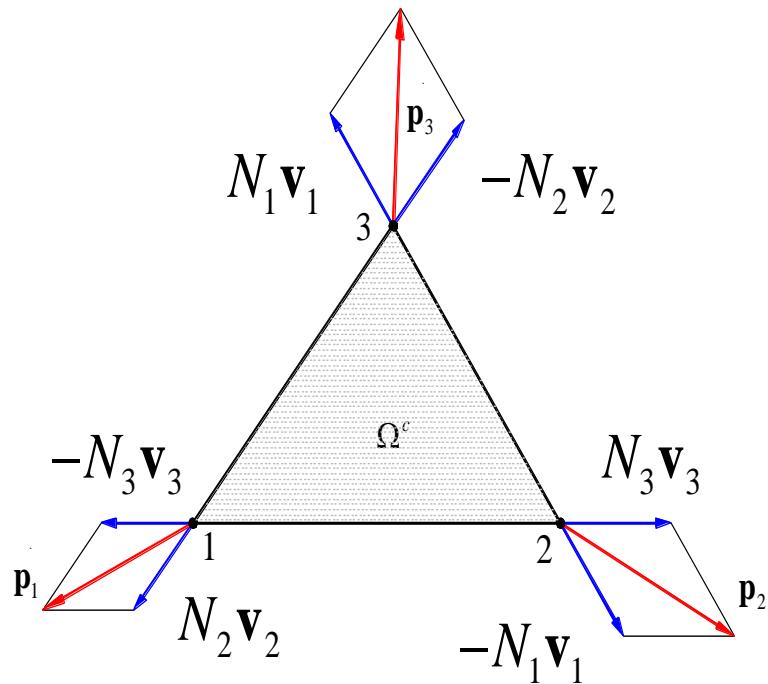
Vector of internal nodal forces:

$$\mathbf{p}^e = \begin{bmatrix} \mathbf{p}_1^e \\ \mathbf{p}_2^e \\ \mathbf{p}_3^e \end{bmatrix} = \frac{t}{2} \begin{bmatrix} \ell_2 \mathbf{p}_2 + \ell_3 \mathbf{p}_3 \\ \ell_1 \mathbf{p}_1 + \ell_3 \mathbf{p}_3 \\ \ell_1 \mathbf{p}_1 + \ell_2 \mathbf{p}_2 \end{bmatrix}$$

# Vector of Natural Forces

The vector of internal forces can be decomposed into components parallel to the element sides:

$$\mathbf{p}^e = \begin{bmatrix} N_2 \mathbf{v}_2^e - N_3 \mathbf{v}_3^e \\ N_3 \mathbf{v}_3^e - N_1 \mathbf{v}_1^e \\ N_1 \mathbf{v}_1^e - N_2 \mathbf{v}_2^e \end{bmatrix}$$



Vector of Natural Forces

$$\mathbf{N} = \begin{bmatrix} N_1 \\ N_2 \\ N_3 \end{bmatrix}$$

# Natural Stresses

Comparing both expression available for  $\mathbf{p}^e$ :

$$\begin{bmatrix} N_2 \mathbf{v}_2^e - N_3 \mathbf{v}_3^e \\ N_3 \mathbf{v}_3^e - N_1 \mathbf{v}_1^e \\ N_1 \mathbf{v}_1^e - N_2 \mathbf{v}_2^e \end{bmatrix} = \frac{t}{2} \begin{bmatrix} \ell_2 \mathbf{\rho}_2 + \ell_3 \mathbf{\rho}_3 \\ \ell_1 \mathbf{\rho}_1 + \ell_3 \mathbf{\rho}_3 \\ \ell_1 \mathbf{\rho}_1 + \ell_2 \mathbf{\rho}_2 \end{bmatrix}$$

We obtain the Vector of Natural Forces, as function of Cauchy Stresses, and we identify some "Natural Stresses" ( $\sigma_1, \sigma_2, \sigma_3$ ):

$$\mathbf{N} = \begin{bmatrix} N_1 \\ N_2 \\ N_3 \end{bmatrix} = \frac{t}{2} \begin{bmatrix} h_1 \left( \frac{\cos \beta}{\sin \gamma \sin \alpha} \sigma_{\hat{y}} - \frac{\sin \beta}{\sin \gamma \sin \alpha} \tau_{\hat{x}\hat{y}} \right) \\ h_2 \left( \frac{\cos \gamma}{\sin \beta \sin \alpha} \sigma_{\hat{y}} + \frac{\sin \gamma}{\sin \beta \sin \alpha} \tau_{\hat{x}\hat{y}} \right) \\ h_3 \left( \sigma_{\hat{x}} - \frac{\cos \beta \cos \gamma}{\sin \beta \sin \gamma} \sigma_{\hat{y}} + \frac{\cos(\beta - \gamma)}{\sin \beta \sin \gamma} \tau_{\hat{x}\hat{y}} \right) \end{bmatrix}$$

$$\mathbf{N} = \begin{bmatrix} N_1 \\ N_2 \\ N_3 \end{bmatrix} = \frac{t}{2} \begin{bmatrix} h_1 \sigma_1 \\ h_2 \sigma_2 \\ h_3 \sigma_3 \end{bmatrix}$$

# Vector of Natural Stresses

We group the Natural Stresses" ( $\sigma_1, \sigma_2, \sigma_3$ ) in a Vector of Natural Stresses:

$$\boldsymbol{\sigma}_n = \begin{bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{bmatrix} = \begin{bmatrix} 0 & \frac{\cos \beta}{\sin \gamma \sin \alpha} & -\frac{\sin \beta}{\sin \gamma \sin \alpha} \\ 0 & \frac{\cos \gamma}{\sin \beta \sin \alpha} & \frac{\sin \gamma}{\sin \beta \sin \alpha} \\ 1 & -\frac{\cos \beta \cos \gamma}{\sin \beta \sin \gamma} & \frac{\sin(\beta - \gamma)}{\sin \beta \sin \gamma} \end{bmatrix} \begin{bmatrix} \sigma_{\hat{x}} \\ \sigma_{\hat{y}} \\ \tau_{\hat{x}\hat{y}} \end{bmatrix}$$

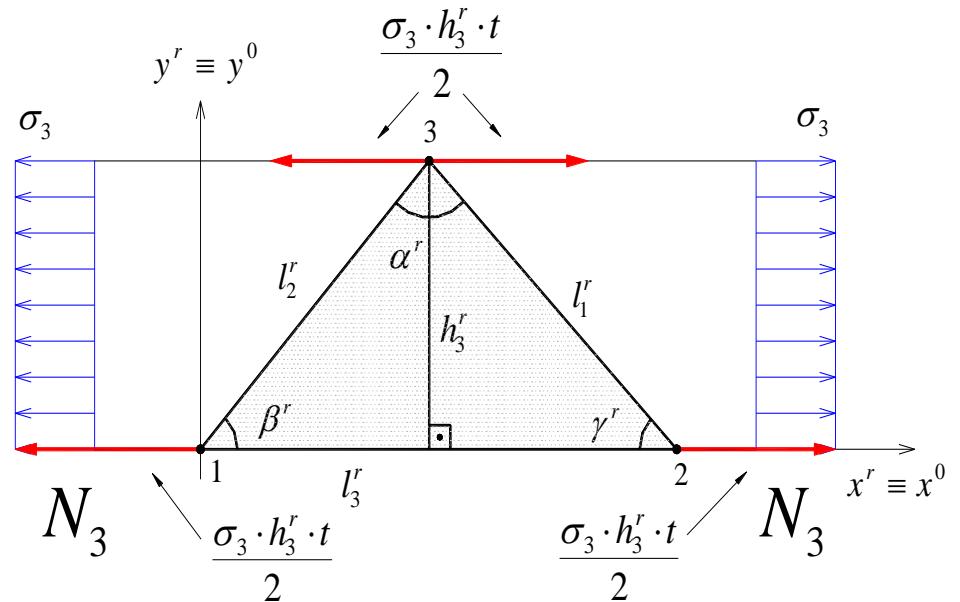
$$\boldsymbol{\sigma}_n = \mathbf{T}^{-T} \hat{\boldsymbol{\sigma}}$$

Exercise 11. Verify the above expression!

# Vector of Natural Stresses

Each natural force  $N_i$  can be understood as the nodal resultant of each natural normal stress field  $\sigma_i$

$$N_i = \frac{t}{2} h_i \sigma_i$$



And since  $h_i = \frac{2A}{\ell_i}$

∴

$$N_i = V \frac{\sigma_i}{\ell_i}$$

# Relationship between the Vectors of Natural Forces and Stresses

In matrix form:

$$\mathbf{N} = V \mathcal{L}^{-1} \boldsymbol{\sigma}_n$$

Length matrix:

$$\mathcal{L} = \begin{bmatrix} \ell_1 & 0 & 0 \\ 0 & \ell_2 & 0 \\ 0 & 0 & \ell_3 \end{bmatrix}$$

$$\boldsymbol{\sigma}_n = \begin{bmatrix} \sigma_1 \\ \sigma_2 \\ \sigma_3 \end{bmatrix}$$

Vector of  
Natural Stresses

# Vector of Natural Deformations

The deformations along the sides of the element are collected in a 'Vector of Natural Deformations':

$$\begin{bmatrix} \varepsilon_1 \\ \varepsilon_2 \\ \varepsilon_3 \end{bmatrix} = \begin{bmatrix} \cos^2 \gamma & \sin^2 \gamma & -\sin \gamma \cos \gamma \\ \cos^2 \beta & \sin^2 \beta & -\sin \beta \cos \beta \\ 1 & 0 & 0 \end{bmatrix} \begin{bmatrix} \varepsilon_{\hat{x}} \\ \varepsilon_{\hat{y}} \\ \gamma_{\hat{x}\hat{y}} \end{bmatrix}$$

Linearized  
Green  
Strains

$$\boldsymbol{\varepsilon}_n = \mathbf{T} \hat{\boldsymbol{\varepsilon}}$$

*Exercise 12. Verify the above expression!*

We remark that  $\boldsymbol{\sigma}_n$  and  $\boldsymbol{\varepsilon}_n$  are energetically conjugate.

Indeed, by the Principle of Virtual Work:

$$\delta \hat{\boldsymbol{\varepsilon}}^T \hat{\boldsymbol{\sigma}} = \delta \boldsymbol{\varepsilon}_n^T \boldsymbol{\sigma}_n , \forall \delta \hat{\boldsymbol{\varepsilon}}$$

$$\delta \hat{\boldsymbol{\varepsilon}}^T \hat{\boldsymbol{\sigma}} = (\mathbf{T} \delta \hat{\boldsymbol{\varepsilon}})^T \boldsymbol{\sigma}_n = \delta \hat{\boldsymbol{\varepsilon}}^T \mathbf{T}^T \boldsymbol{\sigma}_n , \forall \delta \hat{\boldsymbol{\varepsilon}}$$

Thus:  $\boldsymbol{\sigma}_n = \mathbf{T}^{-T} \hat{\boldsymbol{\sigma}}$ , as deduced before.

# Tangent Stiffness Matrix for Argyris' Element

$$\mathbf{k}_t = \frac{\partial \mathbf{g}}{\partial \mathbf{u}} = \mathbf{N}^T \frac{\partial \mathbf{C}}{\partial \mathbf{u}} + \mathbf{C} \frac{\partial \mathbf{N}}{\partial \mathbf{u}} - \frac{\partial \mathbf{f}}{\partial \mathbf{u}}$$

## Geometric Stiffness Matrix

$$\mathbf{k}_g = \mathbf{N}^T \frac{\partial \mathbf{C}}{\partial \mathbf{u}} = \begin{bmatrix} N_2 \frac{\partial \mathbf{v}_2}{\partial \mathbf{u}} - N_3 \frac{\partial \mathbf{v}_3}{\partial \mathbf{u}} \\ -N_1 \frac{\partial \mathbf{v}_1}{\partial \mathbf{u}} + N_3 \frac{\partial \mathbf{v}_3}{\partial \mathbf{u}} \\ N_1 \frac{\partial \mathbf{v}_1}{\partial \mathbf{u}} - N_2 \frac{\partial \mathbf{v}_2}{\partial \mathbf{u}} \end{bmatrix}$$

$$\mathbf{k}_g = \begin{bmatrix} \frac{N_2}{\ell_2} (\mathbf{I}_3 - \mathbf{v}_2 \mathbf{v}_2^T) + \frac{N_3}{\ell_3} (\mathbf{I}_3 - \mathbf{v}_3 \mathbf{v}_3^T) & -\frac{N_3}{\ell_3} (\mathbf{I}_3 - \mathbf{v}_3 \mathbf{v}_3^T) & -\frac{N_2}{\ell_2} (\mathbf{I}_3 - \mathbf{v}_2 \mathbf{v}_2^T) \\ -\frac{N_3}{\ell_3} (\mathbf{I}_3 - \mathbf{v}_3 \mathbf{v}_3^T) & \frac{N_1}{\ell_1} (\mathbf{I}_3 - \mathbf{v}_1 \mathbf{v}_1^T) + \frac{N_3}{\ell_3} (\mathbf{I}_3 - \mathbf{v}_3 \mathbf{v}_3^T) & -\frac{N_1}{\ell_1} (\mathbf{I}_3 - \mathbf{v}_1 \mathbf{v}_1^T) \\ -\frac{N_2}{\ell_2} (\mathbf{I}_3 - \mathbf{v}_2 \mathbf{v}_2^T) & -\frac{N_1}{\ell_1} (\mathbf{I}_3 - \mathbf{v}_1 \mathbf{v}_1^T) & \frac{N_1}{\ell_1} (\mathbf{I}_3 - \mathbf{v}_1 \mathbf{v}_1^T) + \frac{N_2}{\ell_2} (\mathbf{I}_3 - \mathbf{v}_2 \mathbf{v}_2^T) \end{bmatrix}$$

Exact!

# Tangent Stiffness Matrix for Argyris' Element

$$\mathbf{k}_t = \frac{\partial \mathbf{g}}{\partial \mathbf{u}} = \mathbf{N}^T \frac{\partial \mathbf{C}}{\partial \mathbf{u}} + \mathbf{C} \frac{\partial \mathbf{N}}{\partial \mathbf{u}} - \frac{\partial \mathbf{f}}{\partial \mathbf{u}}$$

## External Stiffness Matrix

$$\mathbf{k}_{\text{ext}} = \frac{\partial \mathbf{f}}{\partial \mathbf{u}}$$

External force vector:

$$\mathbf{f} = \mathbf{f}_{\text{weight}} - \mathbf{f}_{\text{wind}} = \frac{V\rho}{3} \begin{bmatrix} \mathbf{I}_3 \\ \mathbf{I}_3 \\ \mathbf{I}_3 \end{bmatrix} \mathbf{g} - \frac{pA}{3} \begin{bmatrix} \mathbf{I}_3 \\ \mathbf{I}_3 \\ \mathbf{I}_3 \end{bmatrix} \mathbf{n}$$

$$\mathbf{k}_{\text{ext}} = \frac{\partial \mathbf{f}_{\text{wind}}}{\partial \mathbf{u}} = \dots = \frac{p}{6} \begin{bmatrix} \Lambda_1 & \Lambda_2 & \Lambda_3 \\ \Lambda_1 & \Lambda_2 & \Lambda_3 \\ \Lambda_1 & \Lambda_2 & \Lambda_3 \end{bmatrix}$$

Exact!

$$\Lambda_i = \text{Skew}(\mathbf{l}_i) = \ell_i \begin{bmatrix} 0 & -v_i^z & v_i^y \\ v_i^z & 0 & -v_i^x \\ -v_i^y & v_i^x & 0 \end{bmatrix}, i=1,2,3$$

Exercise 13. Verify the above expression for  $\mathbf{k}_{\text{ext}}$ !

# Tangent Stiffness Matrix for Argyris' Element

$$\mathbf{k}_t = \frac{\partial \mathbf{g}}{\partial \mathbf{u}} = \mathbf{N}^T \frac{\partial \mathbf{C}}{\partial \mathbf{u}} + \mathbf{C} \frac{\partial \mathbf{N}}{\partial \mathbf{u}} - \frac{\partial \mathbf{f}}{\partial \mathbf{u}}$$

Constitutive Stiffness Matrix

$$\mathbf{k}_c = \mathbf{C} \frac{\partial \mathbf{N}}{\partial \mathbf{u}}$$

*Exact!*

*Defining the vector of Natural Displacements*

$$\mathbf{a} = \begin{bmatrix} \ell_1 - \ell_1^0 \\ \ell_2 - \ell_2^0 \\ \ell_3 - \ell_3^0 \end{bmatrix}$$

*There exist some kind of relationship  $\mathbf{N} = \mathbf{N}(\mathbf{a})$  so that*

$$\mathbf{k}_c = \mathbf{C} \frac{\partial \mathbf{N}}{\partial \mathbf{a}} \frac{\partial \mathbf{a}}{\partial \mathbf{u}} = \mathbf{C} \mathbf{k}_n \mathbf{C}^T$$

$$\mathbf{k}_n = \frac{\partial \mathbf{N}}{\partial \mathbf{a}} \quad \text{'is the } \underline{\text{Natural Stiffness Matrix}'}$$

# Tangent Stiffness Matrix for Argyris' Element

A simplification: Linear elastic material behavior

Thus, a linear relationship  $\mathbf{N} = \mathbf{k}_n^r \mathbf{a}$  exists

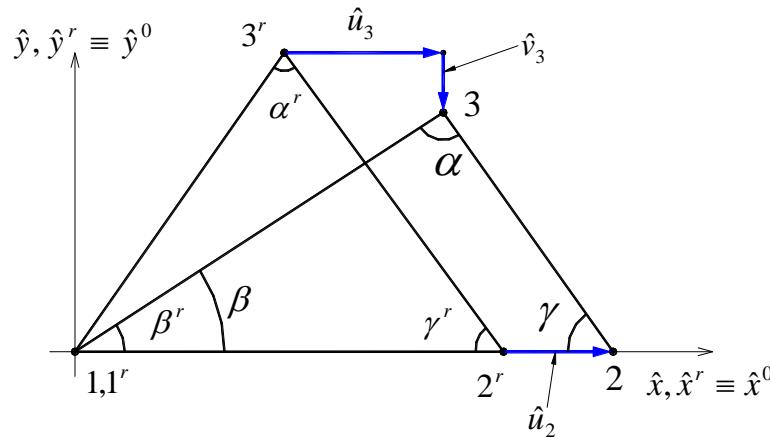
Where  $\mathbf{k}_n^r = \frac{\partial \mathbf{N}}{\partial \mathbf{a}}$  is a  $3 \times 3$  constant natural stiffness matrix

And therefore

$$\mathbf{k}_c = \mathbf{C} \mathbf{k}_n^r \mathbf{C}^T$$

??

# A linear elastic simplification for $K_c$



$$\hat{\boldsymbol{\epsilon}} = \begin{bmatrix} \epsilon_{\hat{x}} \\ \epsilon_{\hat{y}} \\ \gamma_{\hat{x}\hat{y}} \end{bmatrix} = \begin{bmatrix} \frac{\partial \hat{u}}{\partial \hat{x}} \\ \frac{\partial \hat{v}}{\partial \hat{y}} \\ \frac{\partial \hat{v}}{\partial \hat{x}} + \frac{\partial \hat{u}}{\partial \hat{y}} \end{bmatrix} = \begin{bmatrix} \frac{u_2}{x_2^r} \\ \frac{v_3}{y_3^r} \\ \frac{x_2^r u_3 - x_3^r u_2}{x_2^r y_3^r} \end{bmatrix}$$

$$\boldsymbol{\epsilon}_n = \begin{bmatrix} \epsilon_1 \\ \epsilon_2 \\ \epsilon_3 \end{bmatrix} = \begin{bmatrix} \frac{\Delta \ell_1}{\ell_1^r} \\ \frac{\Delta \ell_2}{\ell_2^r} \\ \frac{\Delta \ell_3}{\ell_3^r} \end{bmatrix} = \begin{bmatrix} \frac{1}{\ell_1^r} & 0 & 0 \\ 0 & \frac{1}{\ell_2^r} & 0 \\ 0 & 0 & \frac{1}{\ell_3^r} \end{bmatrix} \begin{bmatrix} \Delta \ell_1 \\ \Delta \ell_2 \\ \Delta \ell_3 \end{bmatrix} = \mathcal{L}_r^{-1} \mathbf{a}$$

}

$$\boxed{\boldsymbol{\epsilon}_n = \mathbf{T}_r \hat{\boldsymbol{\epsilon}}}$$

$$\mathbf{T}_r = \begin{bmatrix} \cos^2 \gamma_r & \sin^2 \gamma_r & -\sin \gamma_r \cos \gamma_r \\ \cos^2 \beta_r & \sin^2 \beta_r & -\sin \beta_r \cos \beta_r \\ 1 & 0 & 0 \end{bmatrix}$$

# A linear elastic simplification for $K_c$

*Hooke's Law:*  $\hat{\boldsymbol{\sigma}} = \begin{bmatrix} \sigma_{\hat{x}} \\ \sigma_{\hat{y}} \\ \tau_{\hat{x}\hat{y}} \end{bmatrix} = \hat{\mathbf{D}}\hat{\boldsymbol{\varepsilon}} + \hat{\boldsymbol{\sigma}}_0$        $\hat{\mathbf{D}} = \frac{E}{1-\nu^2} \begin{bmatrix} 1 & \nu & 0 \\ \nu & 1 & 0 \\ 0 & 0 & \frac{1-\nu}{2} \end{bmatrix}$

*But, now:*  $\boldsymbol{\sigma}_n = \mathbf{T}_r^{-T} \hat{\boldsymbol{\sigma}} = \mathbf{T}_r^{-T} (\hat{\mathbf{D}}\hat{\boldsymbol{\varepsilon}} + \hat{\boldsymbol{\sigma}}_0) = \mathbf{T}_r^{-T} \hat{\mathbf{D}}\hat{\boldsymbol{\varepsilon}} + \mathbf{T}_r^{-T} \hat{\boldsymbol{\sigma}}_0$

$$\boldsymbol{\sigma}_n = \mathbf{T}_r^{-T} \hat{\mathbf{D}} \mathbf{T}_r^{-1} \boldsymbol{\varepsilon}_n + \mathbf{T}_r^{-T} \hat{\boldsymbol{\sigma}}_0$$

*That is*       $\boldsymbol{\sigma}_n = \mathbf{D}_n \boldsymbol{\varepsilon}_n + \boldsymbol{\sigma}_{n0}$

*Where*       $\mathbf{D}_n = \mathbf{T}_r^{-T} \hat{\mathbf{D}} \mathbf{T}_r^{-1}$

# A linear elastic simplification for $K_c$

Recaling the Natural Forces:  $\mathbf{N} = V^r \mathcal{L}_r^{-1} \boldsymbol{\sigma}_n$

$$\mathbf{N} = V^r \mathcal{L}_r^{-1} \mathbf{D}_n \boldsymbol{\varepsilon}_n = V^r \mathcal{L}_r^{-1} \left( \mathbf{T}_r^{-T} \hat{\mathbf{D}} \mathbf{T}_r^{-1} \right) \mathcal{L}_r^{-1} \mathbf{a}$$

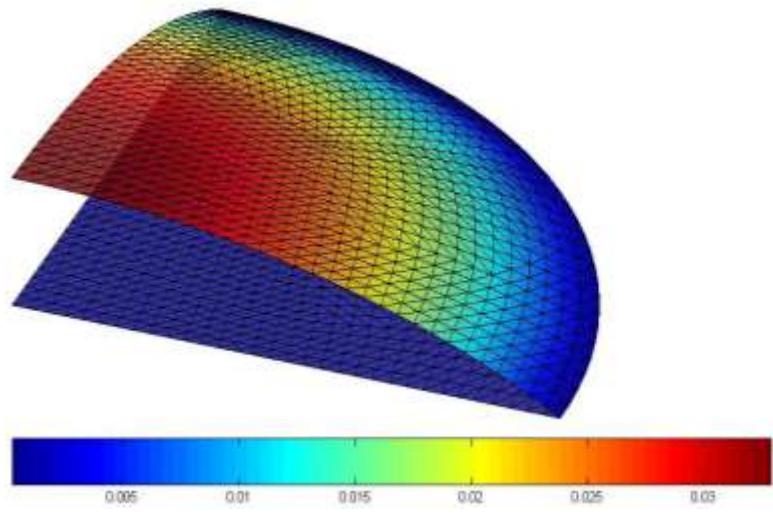
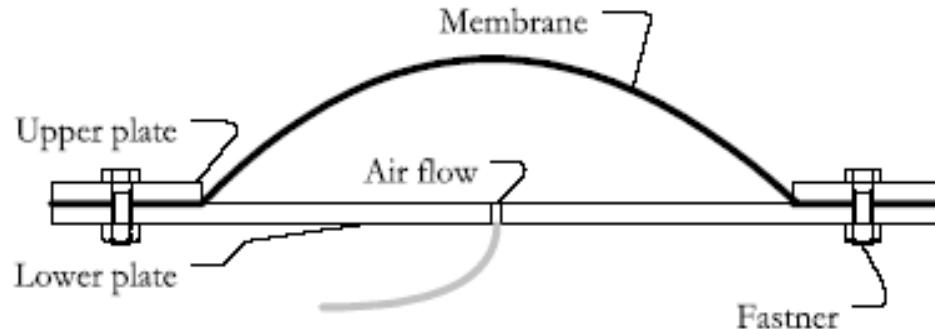
And we arrive to the Natural Stiffness Matrix, (considering small deformations):

$$\mathbf{k}_n^r = \frac{\partial \mathbf{N}}{\partial \mathbf{a}} = V^r \mathcal{L}_r^{-1} \left( \mathbf{T}_r^{-T} \hat{\mathbf{D}} \mathbf{T}_r^{-1} \right) \mathcal{L}_r^{-1}$$

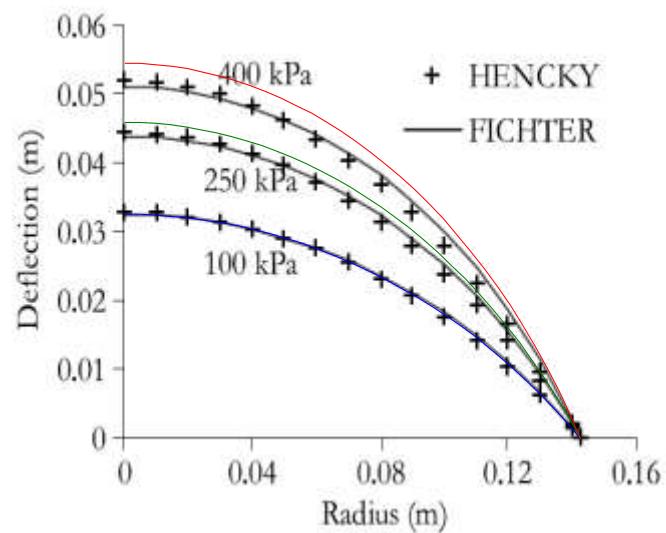
An order 3, symmetric matrix, that can be calculated and stored at the start, and rotated at each Newton's iteration, according to the co-rotational element coordinate system:

$$\mathbf{k}_c = \mathbf{C} \mathbf{k}_n^r \mathbf{C}^T$$

# A benchmark: an axisymmetric pressurized membrane



deformed shape,  
as calculated by SATS



comparison between SATS  
and theoretical results